AEROSPACE VEHICLE DESIGN LABORATORY





Prof. Yee, Kwanjung

E-mail: kjyee@snu.ac.kr Office. 302-627

Lab. 301-1213

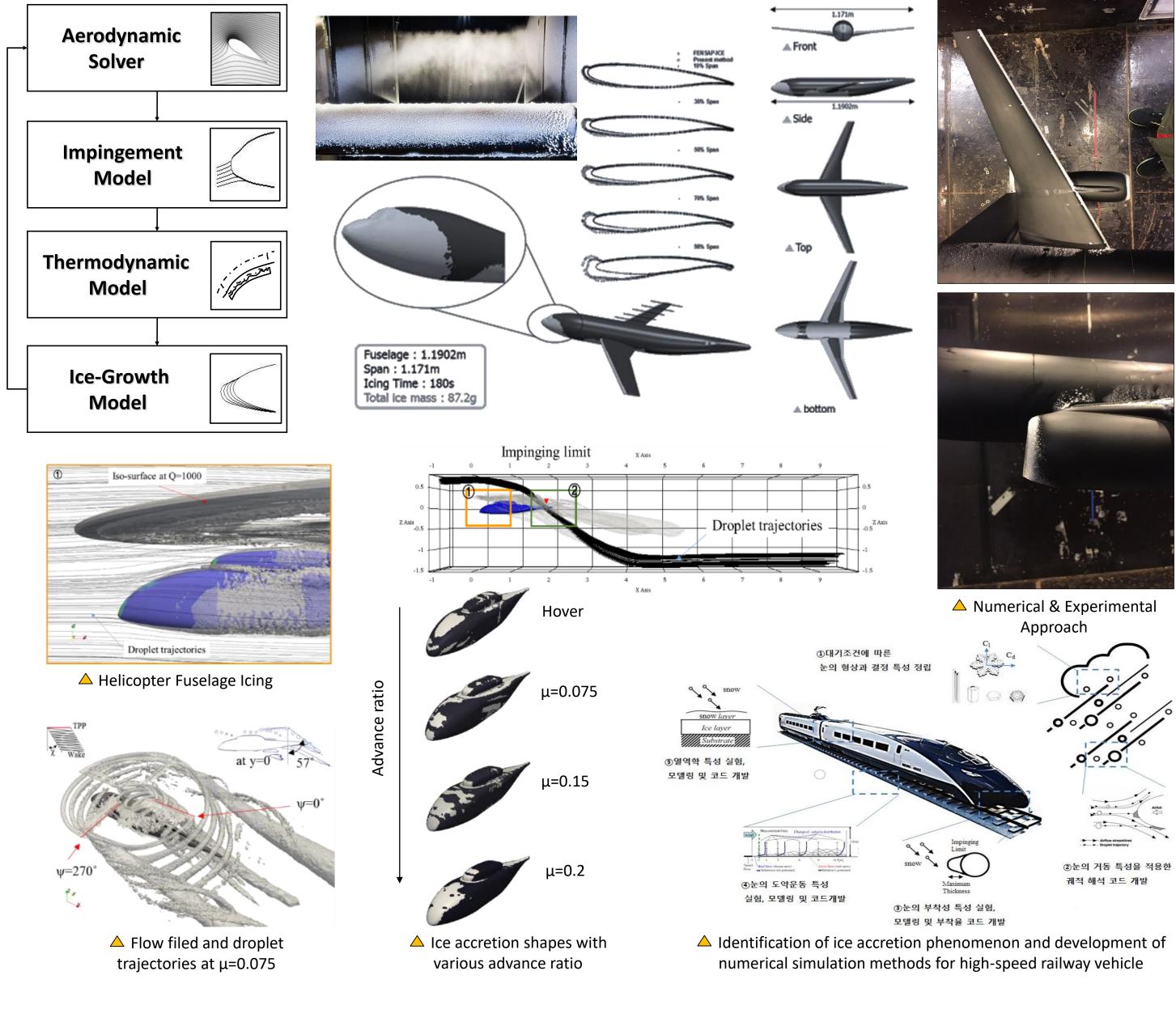
Tel. Office. 02-880-4151

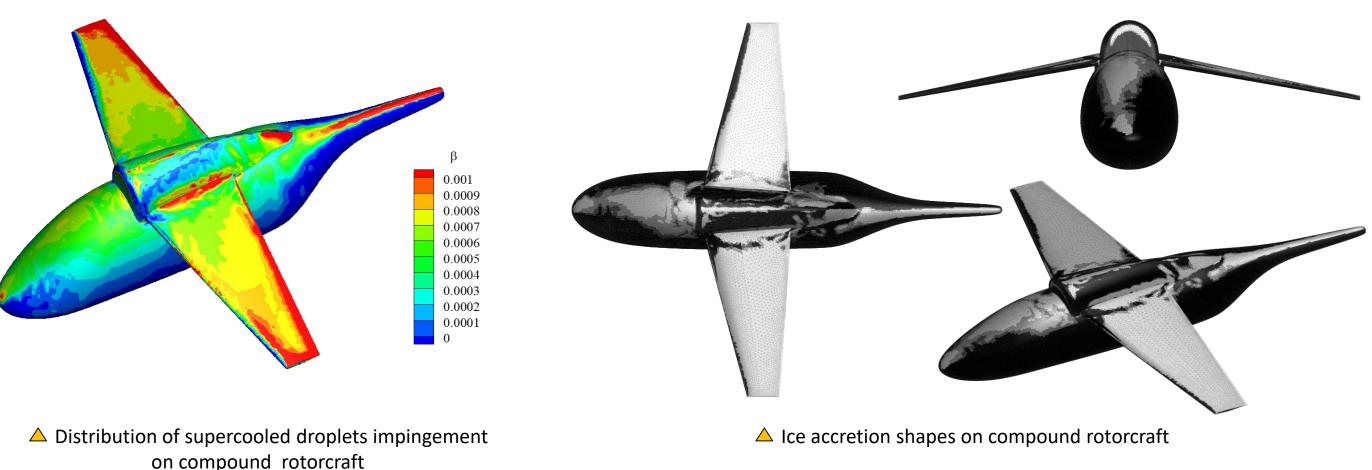
Tel. Lab. 02-880-4152

AVIATION SAFETY

> ICING PROBLEM

Below freezing temperature, super cooled liquid water droplets freeze on the aircraft surface





WAKE HAZARD AVOIDANCE

Two pairs of wake vortex interaction study for increasing airport capacity



△ Wingtip Vortex

Initialization	$h^* < 1.25$ yes	interactions h^*	< 0.75 yes	wake vortex pair		of single wake vortex	- Dissipative	Merging
$\Gamma_{u} \stackrel{Upper}{\longrightarrow} wake\ vortex$ $R_{\Gamma} = \Gamma_{u}/\Gamma_{l}$	h* no		no	coherent vortex _ structure destruction		turbulence dissipation	Most Dissipative	Wrapping
Lower wake vortex		counter-rot. vortexinteractions		upper vortex pair —< linking	$eq.(10)^{\#} < 0 $ yes	lower pair is deformed by ring formation	- Dissipative	Ring deformation
# eq. (10): Δz _{lifes]}	$c_{\mathrm{pan}} - \Delta z_0^* = c_{20}(R_{\Gamma})$	$()^2 + c_{11}R_{\Gamma}h^* + c_{02}(h^*)^2 + c_{12}(h^*)^2 + c_{12}(h^*)^2$	$-c_{10}R_{\Gamma}+c_{01}R_{\Gamma}$	$h^* + c_{00}$		_two pairs are dissipated separately	- Worst Dissipative	Parallel ring dissipation
	Diffusion phase		Non-linear	r interactions	Rapid	decay phase	Final State	Decay processes
//	Snapshots sel	lection rule for each case						S. Carlotte
Hv30Rg07	2. Core-line 3. Start of no	instability on-linear interactions n-linear interactions						
Hv25Rg07	y-axis: vertica	l distance, <i>x-axis</i> : time						
Hv20Rg07	Diffusion Phase						5.3	6.8 <i>t</i> *
Hv15Rg07						Rapid Do	ecay hase	
Hv10Rg07						$\Gamma_0 = 360m^2/s$		
Hv05Rg07						$t^* = 5.3$		
								5.3 <i>t</i> *

ABOUT LABORATORY

LAB MEMBERS

Post Doc : 1 Ph.D. Candidate : 8 Master Course : 9 Researcher : 1

LAB INFORMATION

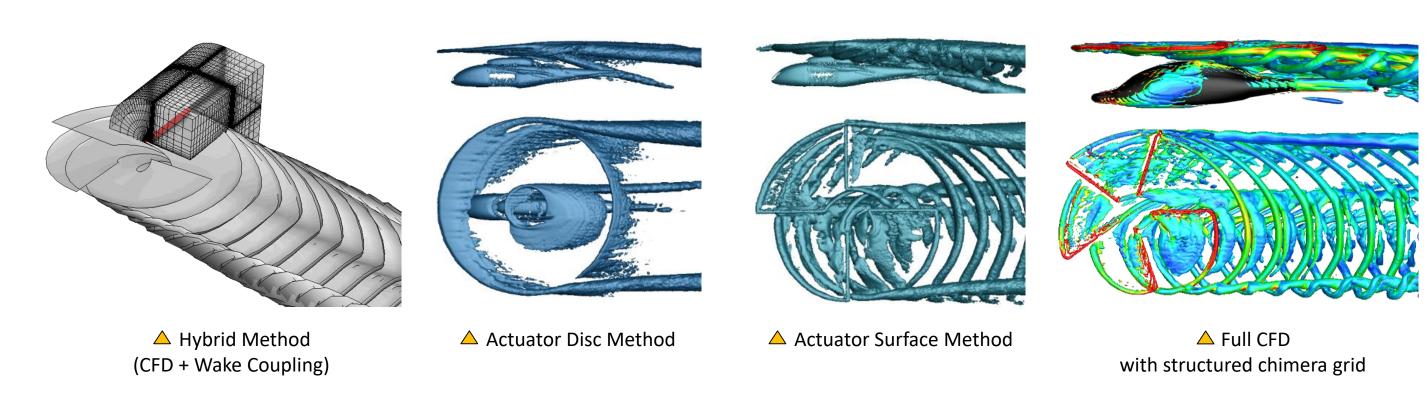
Location: Engineering Building 301-1213

Tel: 02) 880 – 4152 Homepage: avdl.snu.ac.kr

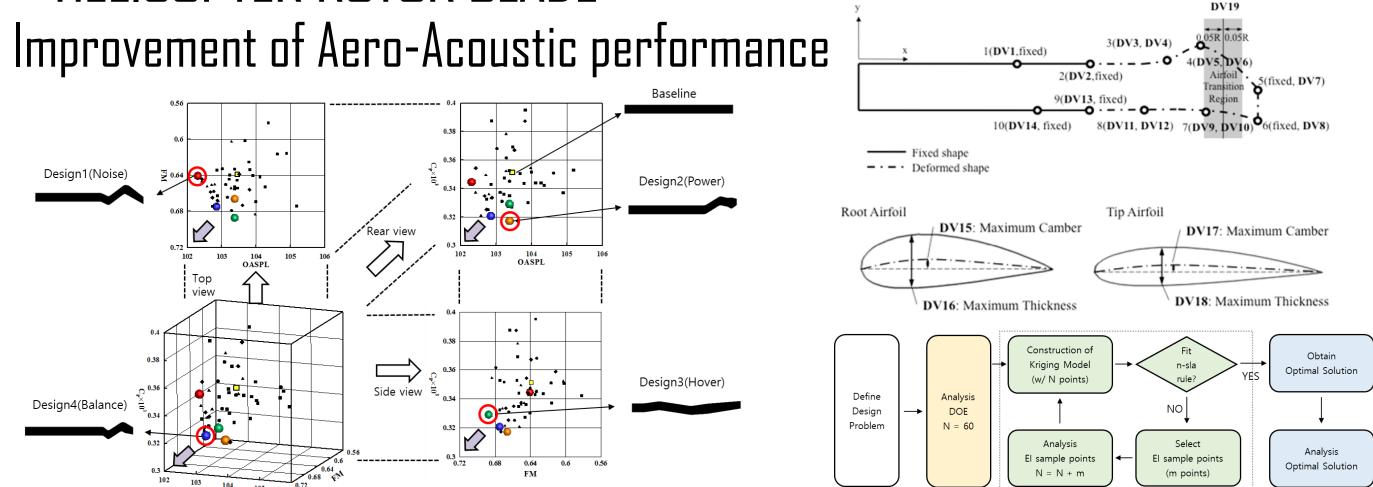
DESIGN OPTIMIZATION

> ROTOR AERODYNAMIC ANALYSIS

Estimating helicopter performance requires large computational time, efficient CFD technique is required



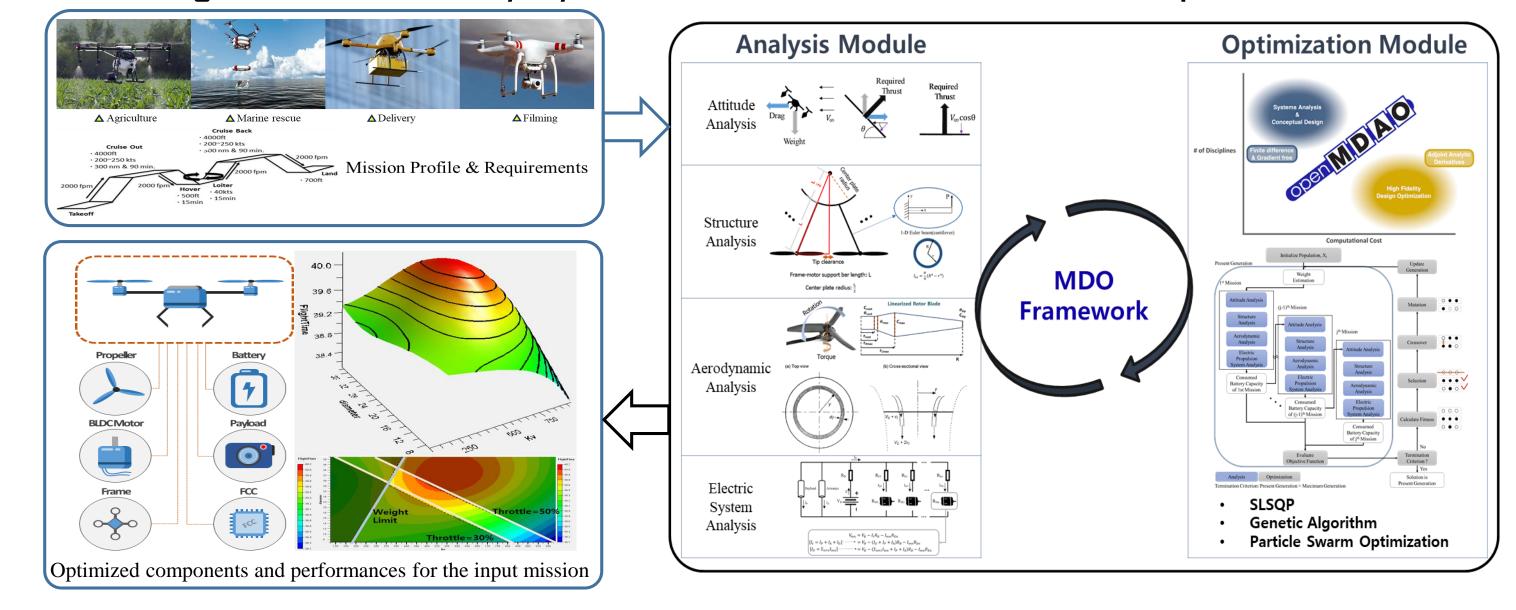
HELICOPTER ROTOR BLADE



MULTI-DISCIPLINARY ROTORCRAFT DESIGN

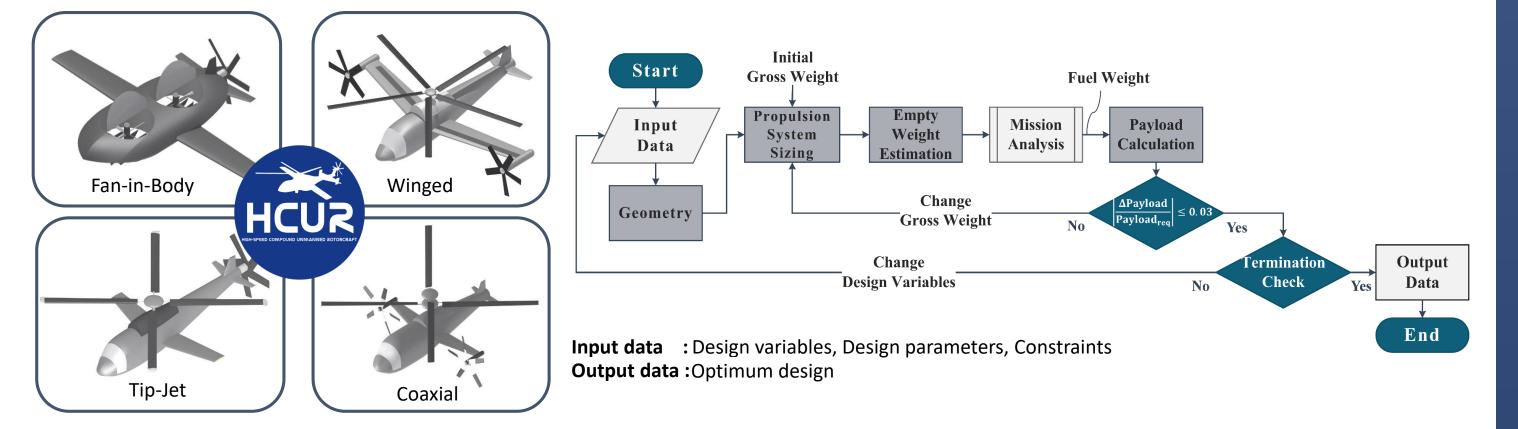
PRELIMINARY ROTORCRAFT DESIGN

Develop an optimization design framework for the required mission Design and test safety system of the multirotor for safe operations



HCUR(HIGH-SPEED COMPOUND UNMANNED ROTORCRAFT)

Conceptual design of compound rotorcraft for future warfare



> VFS 36th annual student design competition (2nd place)

Conceptual design of Extreme Altitude Mountain Rescue Vehicle

